Approximate Trigonometric Series Expansions of

some Bounded Solutions in the Constant Radial

Acceleration Problem

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I. Introduction

In the last decades the studies on the dynamics of a spacecraft that moving in the gravitational field of a celestial body is subjected to some low-thrust propulsion system have attracted the attention of many researchers in Astrodynamics.

Here we will consider a particular case, the so called Tsien problem [1], in which the spacecraft moves in a circular Keplerian orbit (often referred to as the parking orbit) and after a given time $t_0 = 0$ it is subjected to a constant outward radial acceleration.

By using the integrals of energy and angular momentum this problem can be reduced to a two-dimensional scenario and denoting by r and θ the polar coordinates that give the position of the spacecraft in the plane of the orbit [1], the differential equations that define their motion are

$$\frac{d\theta}{dt} = \frac{\sqrt{\mu r_0}}{r^2}, \quad \frac{d^2r}{dt^2} = \frac{\mu}{r^2} \left(\frac{r_0}{r} + \eta \frac{r^2}{r_0^2} - 1 \right) \tag{1}$$

where r_0 is the radius of the parking orbit, μ the gravitational parameter and $\eta \mu/r_0^2$ the dimensionless

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propulsive acceleration ($\eta \geq 0$). The initial conditions at $t_0 = 0$ are

$$r(0) = r_0, \ r'(0) = 0, \ \theta(0) = 0, \ \theta'(0) = \sqrt{\mu/r_0^3}.$$
 (2)

In the early paper of Tsien [1] this author showed that only in the case that the thurst parameter $\eta > 1/8$ the spacecraft escapes from the gravitational field of the celestial body in the sense that it attains a parabolic velocity. The time required to reach this parabolic velocity can be expressed in terms of elliptic functions. For $0 < \eta < 1/8$ the trajectory is bounded and the radius vector r = r(t) oscillates between $r_0 = 1$ and some $r_A > 1$. Now the second equation of (1) can be integrated to give the time in terms of elliptic integrals of first and second kind that depends on r and θ . A complete derivation of these formulas has been given in the book of Battin [2], pp. 408 as well as in the papers [3] and [4].

A remarkable fact is that in most results presented in the literature the time is obtained as a function of the state variables whereas the natural would be the converse dependence. An exception is the paper of Izzo and Biscani [5] in which the state variables are expressed explicitly in terms of an anomaly related with the time by means of Weierstrass functions. Also Akella [6] has shown that the orbit evolution can be expressed in terms of the incomplete elliptic integral of the third kind.

In [7] by using a suitable set of state redundant variables together with a Sundman type timeregularization the authors derive explicit solutions of Tsien problem for all values of η . These analitical solutions have been used by these authors as test problems to compare several orbit propagation method.

Asymptotic solutions of Tsien problem have been derived in [8] and the case of an elliptic parking orbit has been considered in [9].

However, as remarked by Quarta and Mengali in [10] a transparent analytical description of the spacecraft trajectory for all radial thrust acceleration is not available. With the aim to get a more simple description of the motion in the orbital plane they introduce a new variable ρ by

$$\rho = \rho(\theta) = 1 - \frac{r_0}{r(\theta)},\tag{3}$$

Then according to (1)(2) the functions $\rho = \rho(\theta)$ and $\theta = \theta(t)$ of Tsien problem satisfy the nonlinear

set of differential equations

$$\frac{d^2\rho}{d\theta^2} + \rho = \frac{\eta}{(1-\rho)^2}, \quad \frac{d\theta}{dt} = \sqrt{\frac{\mu}{r_0^3}} (1-\rho)^2, \quad \rho \in [0,1), \tag{4}$$

together with the initial conditions

$$\rho(0) = 0, \quad \rho'(0) = \frac{d\rho}{d\theta}(0) = 1, \quad \theta(0) = 0.$$
(5)

Note that the first equation of (4) has the first integral

$$H(\rho', \rho) \equiv \frac{{\rho'}^2}{2} + \frac{{\rho}^2}{2} - \frac{\eta \rho}{(1 - \rho)} = H(\rho'(0), \rho(0))$$
 (6)

and for the Tsien problem $H(\rho'(0), \rho(0)) = 1/2$.

The type of solution of (4)(5) depends on the value of the constant parameter η , so that $\rho = \rho(\theta; \eta)$ but to simplify the notation we will write simply $\rho = \rho(\theta)$.

First of all if $\eta > 1/8$, it follows from the first integral (6) that the solution $\rho = \rho(\theta)$ of (4) is a monotone increasing function of θ for all $\theta \geq 0$ and $\lim_{\theta \to +\infty} \rho(\theta) = 1$, and by (3) $r(t) \to +\infty$ when $t \to +\infty$. Then the trajectory becomes unbounded and the spacecraft escapes from the gravitational field of the attracting body.

If $\eta \in [0, 1/8)$ by using again (6) it can be seen that the corresponding orbit $\rho = \rho(\theta)$ of (4)–(5) oscillates between a minimum $\rho = \rho_P = 0$ and a maximum $\rho = \rho_A$ where ρ_A is the smaller root of the quadratic equation $\rho^2 - \rho + 2\eta = 0$, i.e.

$$\rho_A = \rho_- = \frac{1 - q}{2} \quad \text{with } q = \sqrt{1 - 8\eta} > 0$$
(7)

and it is an even periodic orbit of θ with half–period

$$\theta_A = \int_0^{\rho_A} \frac{\sigma \sqrt{1 - \rho} \, d\rho}{\sqrt{\rho(\rho^2 - \rho + 2\eta)}} = \int_0^{\pi/2} \frac{2\sqrt{2}\sqrt{1 - \rho_A \sin^2 \phi} \, d\phi}{\sqrt{\cos^2 \phi + q(1 + \sin^2 \phi)}}$$
(8)

where the last expression is to be used for the numerical calculation of θ_A for all $\eta \in [0, 1/8)$ to avoid the singularity of the first integral at both ends of the integration interval. Note that according to (8), $\theta_A = \theta_A(\eta)$ and when $\eta \to 1/8$ the half period $\theta_A \to +\infty$.

For $\eta=1/8$ the unique solution $\rho=\rho(\theta)$ of Tsien problem (4)–(5), $\rho=\rho(\theta;1/8)=\rho^*(\theta)$ satisfies $\rho^*(\theta)<1/2$ and it is monotone increasing for all θ , and $\lim_{\theta\to\infty}\rho^*(\theta)=1/2$. Further,

 $\rho(\theta) = 1/2$ for all θ is a circular non-Keplerian orbit of (4) corresponding to $\eta = 1/8$ (see [11]). Hence $\rho^*(\theta)$ is a non periodic solution of Tsien problem that has $\rho = 1/2$ as a limit cycle.

Here we will consider $\rho = \rho(\theta)$ solutions of (4)–(5) with $\eta \in [0, 1/8)$ so that $\rho = \rho(\theta)$ is an analytic and periodic function of θ with period $2\theta_A$. Note that this periodicity of $\rho(\theta)$ with respect to θ does not implies that the orbit of the Tsien problem $\rho = \rho(\theta(t)), \theta = \theta(t)$ is periodic with respect to t. Only if $2\theta_A$ is a rational multiple of 2π the orbit starting from the initial conditions (5) arrives to the same initial point.

Furthermore, since $\rho = \rho(\theta)$ it is an even function of θ it can be expressed as a cosine series Fourier expansion of type

$$\rho(\theta) = \frac{c_0}{2} + \sum_{j=1}^{\infty} c_j \cos\left(\frac{j\pi\theta}{\theta_A}\right)$$
 (9)

with

$$c_j = c_j(\eta) = \frac{1}{\theta_A} \int_0^{\theta_A} \rho(\theta) \cos\left(\frac{j\pi\theta}{\theta_A}\right) d\theta, \quad j = 0, 1, \dots$$
 (10)

Note that as a consequence of Riemann–Lebesgue Lemma [12] if $\rho = \rho(\theta) \in \mathcal{C}^{(m-1)}[0, \theta_A]$ the Fourier coefficients (10) satisfy

$$\lim_{k \to +\infty} k^m \ c_k = 0 \tag{11}$$

i. e. the coefficients of (9) decay very fast with k. In other words the coefficients c_k of high wave numbers corresponding to rapidly oscillating waves must be very small. This holds for all $\eta \in [0, \eta^* = 1/8)$ however when the values of η are close to 1/8 since the half period θ_A tends to ∞ this convergence is slower. Furthermore the N-th partial sums of (9)

$$S_N(\rho(\theta)) = \frac{c_0}{2} + \sum_{j=1}^{N} c_j \cos\left(\frac{j\pi\theta}{\theta_A}\right)$$
 (12)

converge uniformly to $\rho = \rho(\theta)$ in the sense that the following bound holds

$$|\rho(\theta) - S_N(\rho(\theta))| \le K \frac{\ln(N)}{N^m} w(2\pi/N) \quad \theta \in [0, \theta_A]$$
(13)

with a constant K, where w is the modulus of continuity of $\rho^{(m)}(\theta)$. Therefore the uniform convergence of the partial sums is also very fast.

In our case since the solution of (4)–(5) $\rho = \rho(\theta)$ is an analytic function for all $\eta \in [0, \eta^* = 1/8)$ in the interval $[0, \theta_A]$ the above results (11), (13) hold for all non negative integer m and clearly a partial Fourier sum (12) with a few terms, particularly for η not very close to 1/8, gives an accurate representation of the solution of (4)–(5). Note that the smooth function $\rho(\theta)$ changes slowly and the according to (9) the coefficients c_n of high wave numbers corresponding to rapidly oscillating waves must be very small.

Since a direct calculation of the coefficients c_j of (10) only can be carried out numerically after non trivial computations, Quarta and Mengali have proposed in [10] some approximations to (9) that with a good accuracy provide a transparent expression of the orbit and avoid the use of elliptic integrals. In particular the simplest expression $S_1(\theta) = (\rho_A/2) (1 - \cos(\pi\theta/\theta_A))$ satisfies the end conditions $S_1(0) = \rho(0) = 0$ and $S_1(\theta_A) = \rho(\theta_A) = \rho_A$. Also higher order approximations of type $S_n(\theta) = \sum_{j=0}^n b_j \cos(j\pi\theta/\theta_A)$ have been derived in [10] by using a least squares fitting approach.

The aim of this note is to propose an alternative approach to obtain accurate approximations to the solution $\rho = \rho(\theta; \eta)$ of (4)–(5) up to any order. These approximate solutions are obtained as cosine series Hermite interpolating polynomials of $\rho(\theta; \eta)$ that satisfy the differential equation at both ends of the semi–period $[0, \theta_A]$ up to any order. It is shown that such polynomials are very accurate approximations to $\rho(\theta; \eta)$ even for values of the parameter η close to to $\eta^* = 1/8$ when $\theta_A \to +\infty$. Moreover these approximations of $\rho(\theta; \eta)$ together with the second equation of (4) will allow us to obtain a Kepler's type equation relating the polar angle θ with the time.

II. Trigonometrically fitted approximations of the orbit

For a given $\eta \in [0, \eta^* = 1/8)$ and a non negative integer n we consider the cosine trigonometric polynomials

$$T_n(\theta) = T_n(\theta; \eta) = \sum_{j=0}^{2n+1} \beta_j \cos\left(\frac{j\pi\theta}{\theta_A}\right)$$
 (14)

that are intended to approximate the solution $\rho = \rho(\theta; \eta)$ of Eqs. (4) and (5).

Clearly, the form (14) is motivated by the Fourier series expansion of the exact solution although it is not a truncation of this series expansion. To determine the (2n+2) coefficients of (14) we will use Hermite interpolatory conditions at both ends of the θ -interval $[0, \theta_A]$. The initial conditions

(4) imply that $T_n(0) = 0$, $T'_n(0) = 0$ and on the other hand, the symmetry implies that all odd order derivatives of T_n at both ends of the interval $[0, \theta_A]$ vanish, then we must include only even order derivatives at both ends

$$T_n^{(2j)}(0) = \rho^{(2j)}(0) \quad j = 1, \dots, k_1,$$

 $T_n^{(2j)}(\theta_A) = \rho^{(2j)}(\theta_A) \quad j = 1, \dots, k_2,$

$$(15)$$

with k_1 and k_2 such that $k_1 + k_2 = 2n + 2$.

Here we will consider only symmetric interpolants that will be defined by the conditions

$$T_n^{(2j)}(0) = \rho^{(2j)}(0), \quad T_n^{(2j)}(\theta_A) = \rho^{(2j)}(\theta_A), \quad j = 0, \dots, n$$
 (16)

and this trigonometrically fitted interpolant will be referred to as the interpolant of order n.

First of all observe that T_n depends linearly on the (2n+2) coefficients $\beta_j, j = 0, \dots, 2n+1$ and the linear system (15) in these unknowns is non singular and therefore this interpolant is well defined for any $\eta \in [0, \eta^* = 1/8)$.

For the computation of the even order derivatives of $\rho(\theta)$ we use the differential equation (4) so that in the recursive computation of successive derivatives, ρ'' is replaced by $-\rho + \eta(1-\rho)^{-2}$ and then we get the derivatives up to any order as a functions of ρ and ρ' . First of all, according to the initial conditions (3), to obtain these derivatives at the left end $\theta = 0$ we substitute $(\rho, \rho') \to (0, 0)$. In particular for the first orders we get

$$\rho(0) \equiv L_0 = 0,$$

$$\rho^{(2)}(0) \equiv L_2 = \eta,$$

$$\rho^{(4)}(0) \equiv L_4 = 2\eta^2 - \eta,$$

$$\rho^{(6)}(0) \equiv L_6 = 22\eta^3 - 4\eta^2 - \eta,$$

$$\rho^{(8)}(0) \equiv L_8 = \eta (584\eta^3 - 120\eta^2 + 6\eta - 1).$$
(17)

In general $\rho^{(2k)}(0)$ is a polynomial in the thrust parameter η of degree k and integer coefficients.

For the right end $\theta = \theta_A$, similarly substituting $(\rho, \rho') \to (\rho_A, 0)$ we have

$$\rho(\theta_A) \equiv R_0 = \rho_A,
\rho^{(2)}(\theta_A) \equiv R_2 = (\rho_A^2 - 1)/(4(3 - \rho_A)),
\rho^{(4)}(\theta_A) \equiv R_4 = (\rho_A^2 - 1)(5 - 10\rho_A + \rho_A^2)/(4(\rho_A - 3)^3),
\rho^{(6)}(\theta_A) \equiv R_6 = (\rho_A^2 - 1)(11 + 136\rho_A - 146\rho_A^2 - 16\rho_A^3 - \rho_A^4)/(4(\rho_A - 3)^5).$$
(18)

The expressions $\rho^{(2k)}(\theta_A)$ are continuous rational functions in ρ_A .

For the explicit computation of the (2n+2) coefficients $\beta_0, \beta_1, \dots, \beta_{2n+1}$ of $T_n(\theta)$ in the case of symmetric interpolants defined by the condition

$$T_n^{(2j)}(0) = L_{2j}, \quad T_n^{(2j)}(\theta_A) = R_{2j}, \quad j = 0, \dots, n,$$
 (19)

observe that these equations can be written equivalently as two sets of (n + 1) equations in the form

$$\frac{(-1)^j}{2 w^{2j}} \left(T_n^{(2j)}(0) + T_n^{(2j)}(\theta_A) \right) = (-1)^j \left(\frac{L_{2j} + R_{2j}}{2 w^{2j}} \right) \equiv S_{2j}, \quad j = 0, 1, \dots, n$$
 (20)

$$\frac{(-1)^j}{2 w^{2j}} \left(T_n^{(2j)}(0) - T_n^{(2j)}(\theta_A) \right) = (-1)^j \left(\frac{L_{2j} - R_{2j}}{2 w^{2j}} \right) \equiv S_{2j+1}, \tag{21}$$

where $w = \pi/\theta_A$. The set (20) defines the (n+1) coefficients $\beta_0, \beta_2, \dots, \beta_{2n}$ by the linear system

$$\beta_0 + \beta_2 + \dots + \beta_{2n} = S_0$$

$$\beta_2 (2^2) + \dots + \beta_{2n} (2n)^2 = S_2$$

$$\dots \dots \dots \dots$$

$$\beta_2 2^{(2n)} + \dots + \beta_{2n} (2n)^{(2n)} = S_{2n}$$
(22)

whereas the set (21) defines the (n+1) coefficients $\beta_1, \beta_3, \dots, \beta_{2n+1}$ by the linear system

$$\beta_{1} + \beta_{3} + \ldots + \beta_{2n+1} = S_{1}$$

$$\beta_{1} (1^{3}) + \beta_{3} (3^{3}) + \ldots + \beta_{2n+1} (2n+1)^{2} = S_{3}$$

$$\ldots$$

$$\beta_{1} 1^{(2n)} + \beta_{3} (3)^{(2n)} \ldots + \beta_{2n+1} (2n+1)^{(2n)} = S_{2n+1}$$

$$(23)$$

Clearly, Eq. (22) is a non singular linear system in the unknowns even coefficients $\beta_0, \beta_2, \dots \beta_{2n}$ and similarly for (23) in the odd coefficients. Hence, for all integer $n \geq 0$ there is a uniquely defined symmetric interpolant T_n with coefficients depending on ρ_A and η .

For the first order approximation

$$T_1(\theta) = \sum_{j=0}^{3} \beta_j \cos(jw\theta), \tag{24}$$

with $w = \pi/\theta_A$, we have the system

$$\beta_{0} + \beta_{2} = S_{0}$$

$$\beta_{2}(2^{2}) = S_{2}$$

$$\beta_{1} + \beta_{3} = S_{1},$$

$$\beta_{1} (1^{2}) + \beta_{3} (3^{2}) = S_{3}$$

$$(25)$$

with

$$S_0 = (L_0 + R_0)/2,$$
 $S_1 = (L_0 - R_0)/2,$ $S_2 = (-1)(L_2 + R_2)/(2 w^2),$ $S_3 = (-1)(L_2 - R_2)/(2 w^2).$

Then the coefficients β_i are

$$\begin{split} \beta_0 &= -\frac{(\sigma-1)\left[(\sigma-1)^2\theta_A^2 + 16\pi^2(\sigma+1)\right]}{64\pi^2(\sigma+1)} \\ \beta_2 &= -\frac{(\sigma-1)\left(\left(\sigma^2 + 6\sigma + 1\right)\theta_A^2 - 36\pi^2(\sigma+1)\right)}{128\pi^2(\sigma+1)} \\ \beta_1 &= \frac{(\sigma-1)^3\theta_A^2}{64\pi^2(\sigma+1)} \\ \beta_3 &= \frac{(\sigma-1)\left[\left(\sigma^2 + 6\sigma + 1\right)\theta_A^2 - 4\pi^2(\sigma+1)\right]}{128\pi^2(\sigma+1)} \end{split}$$

For the sake of completeness we compare the above trigonometrically fitted approximation $T_1(\theta)$ with the least squares approximation $QM_2(\theta)$ given by Quarta and Mengali in [10] for the values $\eta = 1/80$, $\eta = 1/16$ and $\eta = 19/160$. This approximation is given by

$$QM_2(\theta) = \rho_A \left[\frac{1}{2} \left(1 - \cos \left(\frac{\pi \theta}{\hat{\theta}_A} \right) \right) + b_2 \left(\cos \left(\frac{2\pi \theta}{\hat{\theta}_A} \right) - 1 \right) \right]$$

where b_2 , $\hat{\theta}$ are obtained from Table 1 for several values of η .

Table 1 Coefficients of b_2 , $\hat{\theta}$ and η of $QM_2(\theta)$

$$\begin{array}{c|ccccc} \eta & b_2 & \hat{\theta}_A \\ \hline 1/80 & -4.3357 \times 10^{-5} & 1.01325602218917 \\ 1/16 & -1.8657 \times 10^{-3} & 1.09017029950805 \\ \hline 19/160 & -3.1444 \times 10^{-2} & 1.42575249853646 \\ \hline \end{array}$$

In Figures 1–2, 3–4 and 5–6 we display the defects of the differential equation (4)

$$\delta_d(\theta) = T_1^{(2)}(\theta) + T_1(\theta) - \eta (1 - T_1(\theta))^{-1}$$
(26)

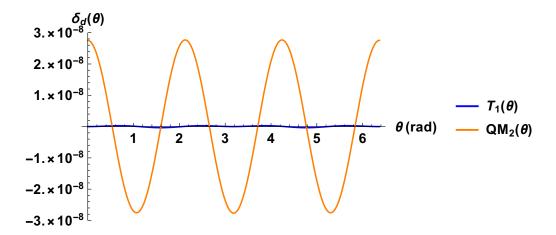


Fig. 1 Defects of the differential equation for the first order interpolant $T_1(\theta)$ and the approximation $QM_2(\theta)$ given by Quarta and Mengali for $\eta = 1/80, \ \eta^* = 1/10$.

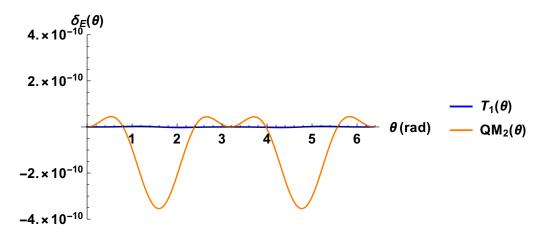


Fig. 2 Defects of the energy integral for the first order interpolant $T_1(\theta)$ and the approximation $QM_2(\theta)$ given by Quarta and Mengali for $\eta=1/80,\ \eta^*=1/10$.

and the first integral of the energy (4)

$$E(\rho(\theta)) \equiv \rho'(\theta)^2 + \rho(\theta)^2 - 2\eta (1 - \rho(\theta))^{-1} = \text{constant}$$
(27)

given by

$$\delta_E(\theta) = E(T_1(\theta)) - E(T_1(0)) \tag{28}$$

for the values of $\eta=1/80,\,1/16$ and 19/160 respectively.

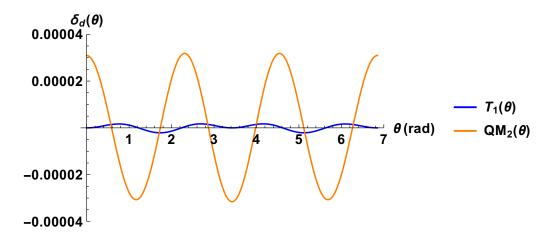


Fig. 3 Defects of the differential equation for the first order interpolant $T_1(\theta)$ and the approximation $QM_2(\theta)$ given by Quarta and Mengali for $\eta = 1/16$, $\eta^* = 1/2$.

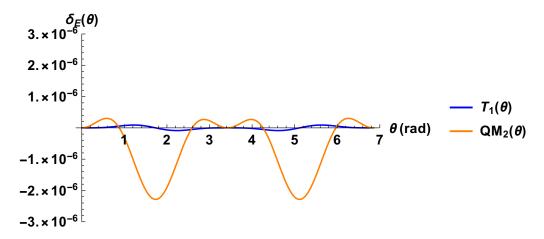


Fig. 4 Defects of the energy integral for the first order interpolant $T_1(\theta)$ and the approximation $QM_2(\theta)$ given by Quarta and Mengali for $\eta=1/16,\ \eta^*=1/2$.

For the second order approximation

$$T_2(\theta) = \sum_{j=0}^{5} \beta_j \cos(jw\theta), \tag{29}$$

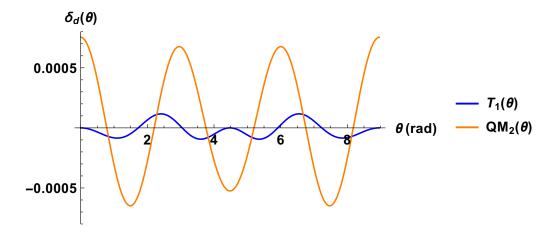


Fig. 5 Defects of the differential equation for the first order interpolant $T_1(\theta)$ and the approximation $QM_2(\theta)$ given by Quarta and Mengali for $\eta = 19/160$, $\eta^* = 95/100$.

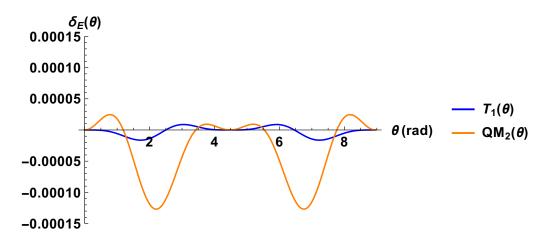


Fig. 6 Defects of the energy integral for the first order interpolant $T_1(\theta)$ and the approximation $QM_2(\theta)$ given by Quarta and Mengali for $\eta = 19/160, \ \eta^* = 95/100$.

with $w=\pi/\theta_A$, the conditions (22),(23) to determine the coefficients β_j of (29) are

$$\beta_0 + \beta_2 + \beta_4 = S_0$$

$$\beta_2 2^2 + \beta_4 4^2 = S_2$$

$$\beta_2 2^4 + \beta_4 4^4 = S_4$$

$$\beta_1 + \beta_3 + \beta_5 = S_1$$

$$\beta_1 + \beta_3 3^2 + \beta_5 5^2 = S_3$$

$$\beta_1 + \beta_3 3^4 + \beta_5 5^4 = S_5$$

that possess the solution

$$\beta_0 = \frac{1}{64}(64S_0 - 20S_2 + S_4), \qquad \beta_1 = \frac{1}{192}(225S_1 - 34S_3 + S_5), \quad \beta_2 = \frac{1}{48}(16S_2 - S_4),$$

$$\beta_3 = \frac{1}{128}(-25S_1 + 26S_3 - S_5), \quad \beta_4 = \frac{1}{192}(-4S_2 + S_4), \qquad \beta_5 = \frac{1}{384}(9S_1 - 10S_3 + S_5).$$

As above, we compare the trigonometrically fitted approximation $T_2(\theta)$ with the least squares approximation $QM_3(\theta)$ given by Quarta and Mengali in [10]

$$QM_3(\theta) = \rho_A \left[\frac{1}{2} \left(1 - \cos \left(\frac{\pi \theta}{\hat{\theta}_A} \right) \right) + \tilde{b}_2 \left(\cos \left(\frac{2\pi \theta}{\hat{\theta}_A} \right) - 1 \right) + \tilde{b}_3 \left(\cos \left(\frac{3\pi \theta}{\hat{\theta}_A} \right) - \cos \left(\frac{\pi \theta}{\hat{\theta}_A} \right) \right) \right]$$

for $\eta=1/80,\,\eta=1/16$ and $\eta=19/160$ given in Table 2

Table 2 Coefficients of \tilde{b}_i , $\hat{\theta}$ and η of $QM_3(\theta)$

η	$ ilde{b}_2$	$ ilde{b}_3$	$\hat{ heta}_A$
1/80	-4.3356×10^{-5}	1.3805×10^{-7}	1.01325602218917
1/16	-1.8640×10^{-3}	3.1705×10^{-5}	1.09017029950805
19/160	-3.1375×10^{-2}	4.2249×10^{-4}	1.42575249853646

A similar comparison can be carried out for other values of η with the coefficients b_j provided by the Table 1 of [10].

In Figures 7–8, 9–10 and 11–12 we display the defects of the differential equation and the the energy integral of $T_2(\theta)$ and $QM_3(\theta)$ for the values of $\eta = 1/80$, 1/16 and 19/160 respectively. Moreover we give here the values of the coefficients of the trigonometrical polynomial for $\eta = 1/16$ that show their rapid decrease

$$\beta_0 = 0.0734963,$$
 $\beta_1 = -0.0732279,$ $\beta_2 = -0.000272905,$ $\beta_3 = 4.6349610^{-6},$ $\beta_4 = -1.0441110^{-7},$ $\beta_5 = 2.5583810^{-9}.$

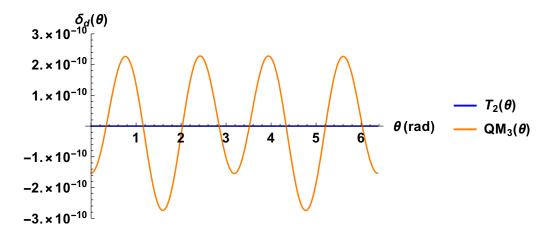


Fig. 7 Defects of the differential equation for the first order interpolant $T_2(\theta)$ and the approximation $QM_3(\theta)$ given by Quarta and Mengali for $\eta = 1/80$, $\eta^* = 1/10$.

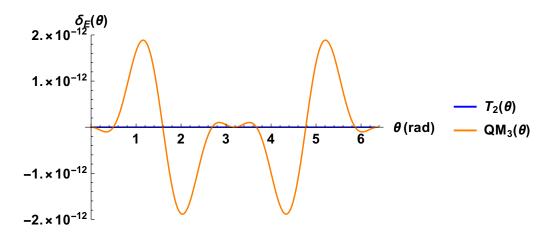


Fig. 8 Defects of the energy integral for the first order interpolant $T_2(\theta)$ and the approximation $QM_3(\theta)$ given by Quarta and Mengali for $\eta = 1/80, \ \eta^* = 1/10$.

As remarked above when the value of the thrust parameter η closes to the critical value $\eta^* = 1/8$ the accuracy of $T_2(\theta)$ decreases. In Figures 11–12 we display the defect(θ) and energy(θ) for $\eta = 0.95$, $\eta^* = 19/160$, and also the values of the the coefficients of the corresponding trigonometrical polynomial $T_2(\theta)$ that do not show their rapid decrease

$$\begin{split} \beta_0 &= 0.206275, \qquad \beta_1 = -0.194264, \qquad \beta_2 = -0.0121868, \\ \beta_3 &= 0.000166625, \ \beta_4 = 9.65595 \times 10^{-6}, \ \beta_5 = -1.39145 \times 10^{-6}. \end{split}$$

The conclusion is that the errors of $QM_3(\theta)$ are about the same order that the $T_1(\theta)$ although

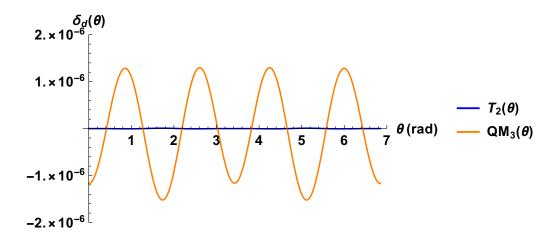


Fig. 9 Defects of the differential equation for the first order interpolant $T_2(\theta)$ and the approximation $QM_3(\theta)$ given by Quarta and Mengali for $\eta = 1/16$, $\eta^* = 1/16$.

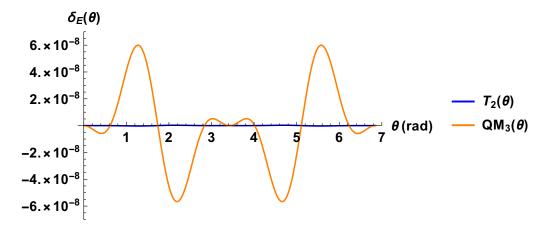


Fig. 10 Defects of the energy integral for the first order interpolant $T_2(\theta)$ and the approximation $QM_3(\theta)$ given by Quarta and Mengali for $\eta=1/16, \eta^*=1/16$.

the adjustment with a least squares procedure makes it slightly better. On the other hand $T_2(\theta)$ is clearly superior to $QM_3(\theta)$ as can be seen in Figures 13–14 for $\eta = 19/160$. Similar behaviour occurs with other values of η . In any case it is worth to remark that there is an analytical expression for the coefficients of $T_1(\theta)$ that makes it available for any $\eta \in [0, 1/8)$ whereas in the case of $QM_3(\theta)$ these must be derived for any particular value of η with least squares approach.

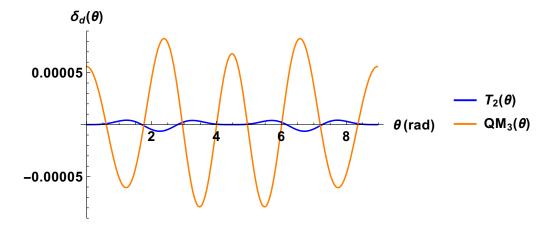


Fig. 11 Defects of the differential equation for the first order interpolant $T_2(\theta)$ and the approximation $QM_3(\theta)$ given by Quarta and Mengali for $\eta = 19/160$, $\eta^* = 95/100$.

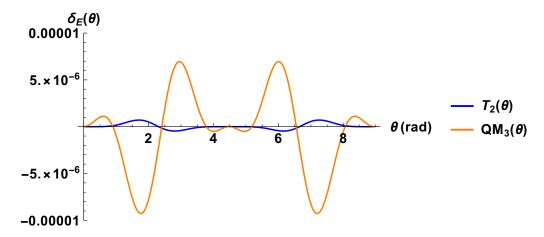


Fig. 12 Defects of the energy integral for the first order interpolant $T_2(\theta)$ and the approximation $QM_3(\theta)$ given by Quarta and Mengali for $\eta = 19/160$, $\eta^* = 95/100$.

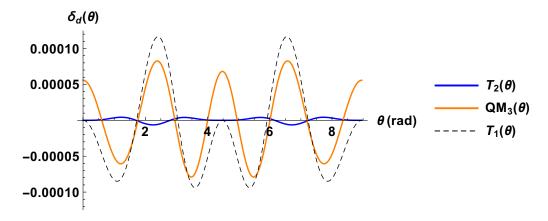


Fig. 13 Defects of the differential equation for the first order interpolants $T_1(\theta)$, $T_2(\theta)$ and the approximation $QM_3(\theta)$ given by Quarta and Mengali for $\eta = 19/160$, $\eta^* = 95/100$.

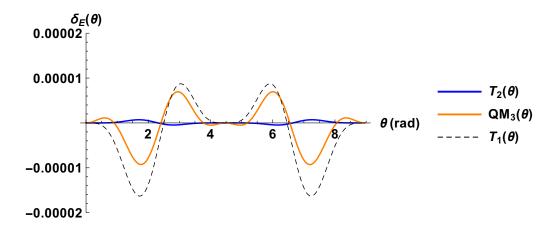


Fig. 14 Defects of the energy integral for the first order interpolants $T_1(\theta)$, $T_2(\theta)$ and the approximation $QM_3(\theta)$ given by Quarta and Mengali for $\eta = 19/160$, $\eta^* = 95/100$.

III. Time approximation

After obtaining a trigonometrical polynomial $T_n(\theta)$ approximating the orbit $\rho = \rho(\theta)$ we need to relate the polar angle θ with the physical time t. From the angular momentum integral we have

$$\frac{d\theta}{dt} = \frac{\sqrt{\mu r_0}}{r^2} = \sqrt{\frac{\mu}{r_0^3}} (1 - \rho)^2 = \sqrt{\frac{\mu}{r_0^3}} \frac{\eta}{\rho'' + \rho},$$

hence,

$$(\rho'' + \rho) d\theta = \eta \sqrt{\frac{\mu}{r_0^3}} dt,$$

and by integration, the time t to reach a given θ satisfies

$$\eta \sqrt{\frac{\mu}{r_0^3}} t = \rho'(\theta) + \int_0^\theta \rho(\theta) d\theta.$$

In the case of periodic orbits when $\rho \simeq T_n$, the right hand side can be approximated by

$$I_n(\theta) = T'_n(\theta) + \int_0^{\theta} T_n(\theta) d\theta = \beta_0 \theta + \sum_{j=1}^{2n+1} \beta_j \left(\frac{1 - j^2 w^2}{jw} \right) \sin(jw\theta),$$

where $\beta_j = \beta_j(\eta)$ are trigonometrically fitted coefficients of T_n and the relation

$$\eta \sqrt{\frac{\mu}{r_0^3}} \ t = I_n(\theta)$$

can be considered as a Kepler type equation to determine θ for each value of t.

Note that for $\theta = \theta_A$, since $w = \pi/\theta_A$, $I_n(\theta_A) = \beta_0 \theta_A$ and

$$t_A = \frac{\beta_0 \theta_A}{\eta} \left(\frac{\mu}{r_0^3}\right)^{-1/2}.$$

IV. Conclusions

A new technique has been proposed to derive trigonometrically fitted approximations to the periodic solutions in the constant, outward radial acceleration problem sometimes referred to as Tsien problem [1]. It can be considered as an alternative solution to the one given by Quarta and Mengali in [10]. A remarkable property of these periodic orbits is that the function $\rho(\theta)$ $1-r(0)/r(\theta)$ is an even periodic solution of a nonlinear second order nonlinear analytical differential equation, and therefore the coefficients of their Fourier series have an spectral convergence to zero. This implies that our trigonometric polynomials $T_n(\theta)$ that mimic the Fourier expansion up to any order share a high accuracy that is uniform in $\theta > 0$. The coefficients of $T_n(\theta) \simeq \rho(\theta)$ are computed by Hermite interpolation up of the solution of the differential equation at both ends of a semiperiod. Taking into account the differential equation this process can be derived recursively up to any order. Two criteria have been proposed to test the quality of these approximate solutions: the defect of differential equation satisfied by $\rho(\theta)$ and also a first integral of this equation and results of some numerical experiments are presented to show the quality of these approximations in the first orders. Further, by using $T_n(\theta)$ an approximation to the relation of the time t and the polar angle $t=t(\theta)$ is obtained. Such a relation can be viewed as a Kepler's type equation. In conclusion the proposed solution provides an approximate representation of the periodic solutions of $\rho = \rho(\theta)$ as a trigonometric polynomial with coefficients easily calculable with a good accuracy that is uniform for all θ .

Acknowledgments

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