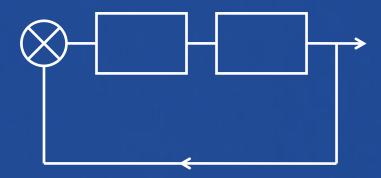




# DEVELOPMENT OF AUTOMATIC FLIGHT CONTROL SYSTEM



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**Course: Avionics** 

Supervisor: Dr. Zoubir

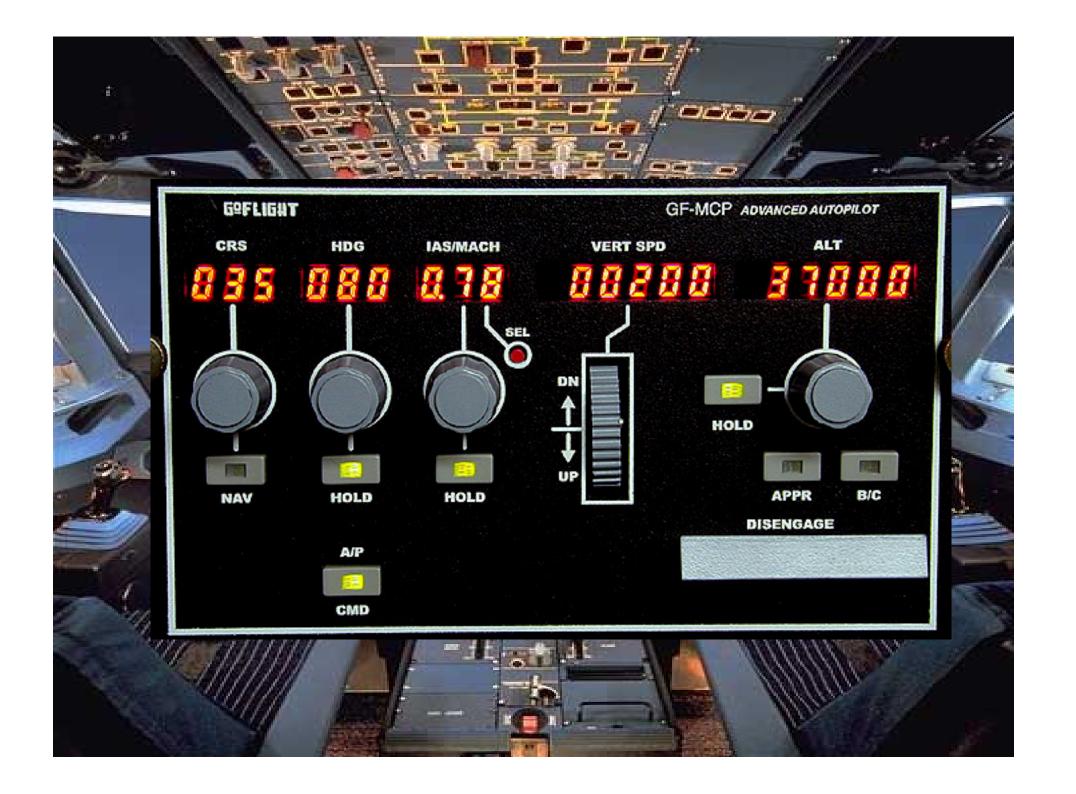


#### Aim:

To develop an autopilot or an AFCS. The Design has been conducted in the Matlab environment .

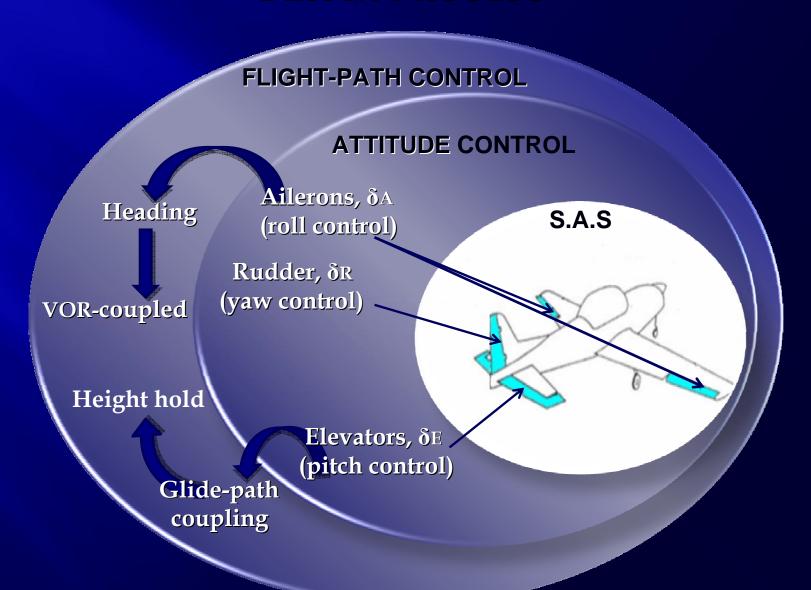
### **Objectives:**

- Know concepts about control system designs.
- Develop flight dynamic models for an aircraft perturbation motion.
- Design state space and transfer function models(longitudinal and lateral).
- Research flying quality analysis.
- Develop and enhance handling and flying qualities.
- Understand and choose which control system design approach is better.
- Be familiar with advance programming techniques using Matlab.





#### **DESIGN PROCESS**





#### **PITCH ATTITUDE**

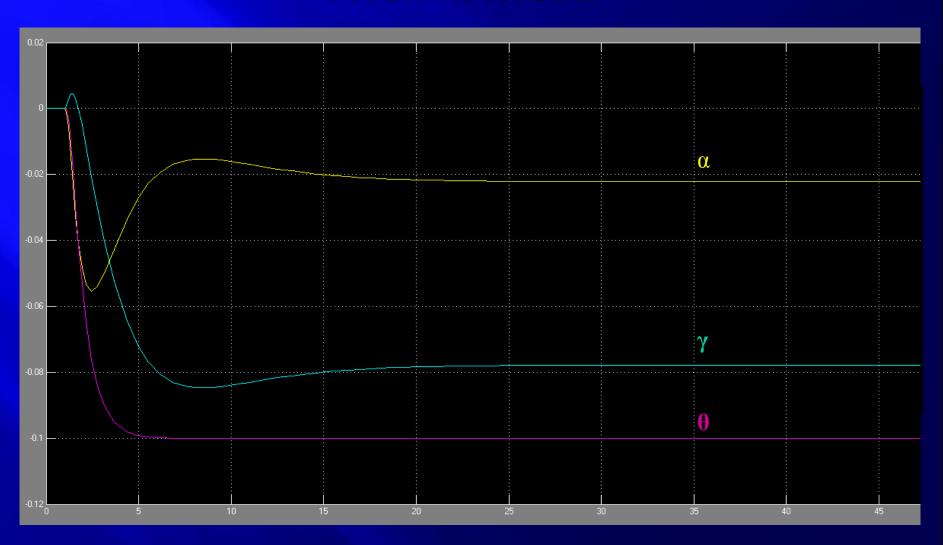
#### PITCH CONTROL



- Longitudinal parameters for State Space model (CHARLIE-1).
- Damping controlled by feedbacks  $\delta E$  ( Kq ,  $K\theta$  ).
- Analyse the flying quality: criterion and analysis.



#### **PITCH ATTITUDE**





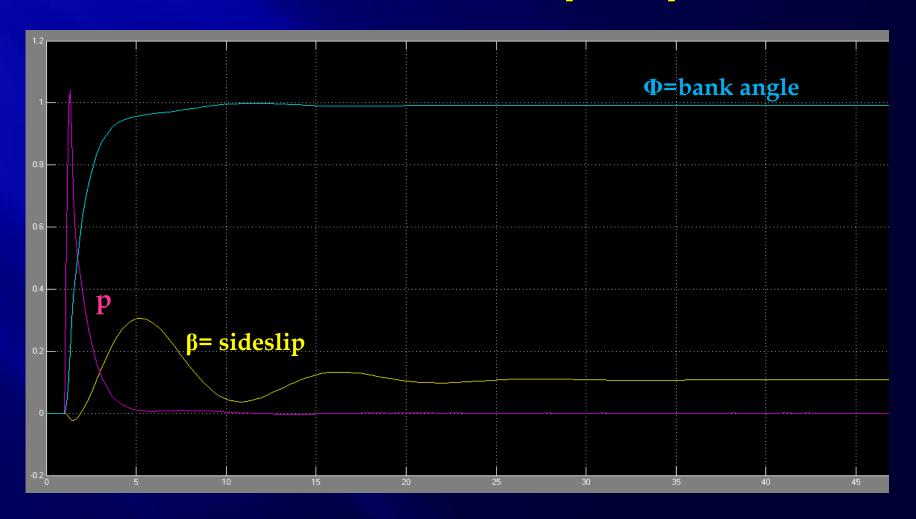
#### **ROLL CONTROL**



- Lateral parameters for State Space model (CHARLIE-1).
- Choose a control system design: Bank angle control with roll rate inner loop damper.
- Damping controlled by feedbacks:  $\delta$ A ( Kq,Kc1,Kc2 , K $\Phi$  ).
- Analyse the flying quality: criterion and analysis.



# Bank angle control system With roll rate inner loop damper





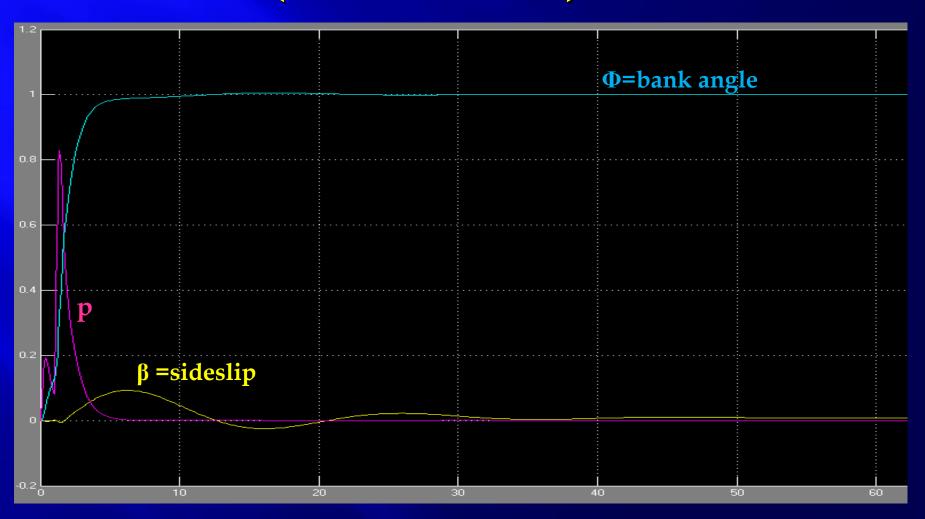
#### YAW CONTROL



- Lateral parameters for State Space model (CHARLIE-1).
- Choose a control system design: ARI (aileron-to-rudder), to control:
- Roll rate.
- -The sideslip (yawing moment).
  - -Control Crossfeed
- $\delta$ A +  $\delta$ R (Kq, Kc1, Kc2, K $\theta$ ).
- Analyse the flying quality: criterion and analysis.

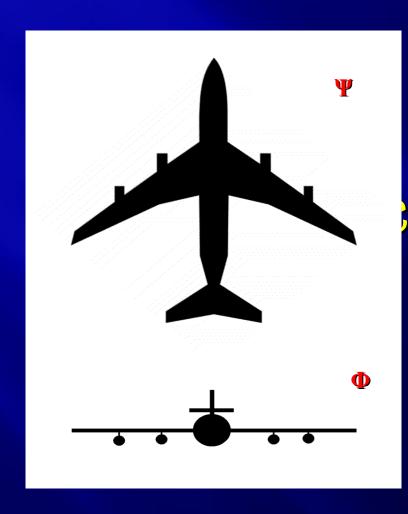


# ARI SYSTEM (aileron-to-rudder)





#### **HEADING (Directional systems)**



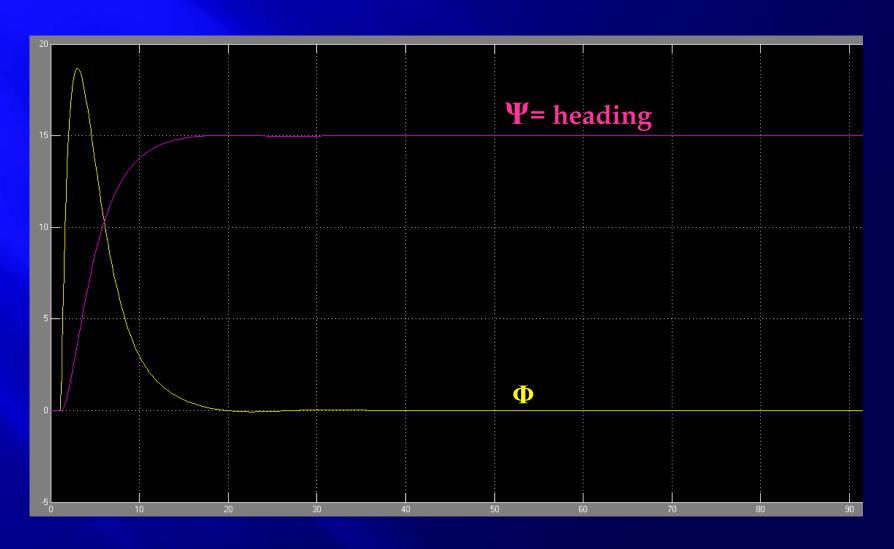
- Ψ = yaw + coordinated turn.
   Φ = bank angle.
- Lateral parameters for State Space

# ONTROCHARUS TEMS

- Choose a control system design : ARI (aileron-to-rudder), as a inner loop.
- $\bullet$   $\delta$ A,  $\delta$ R (Kq, Kc1, Kc2, K $\theta$ , K $\Psi$ ).
- Analyse the flying quality: criterion and analysis.

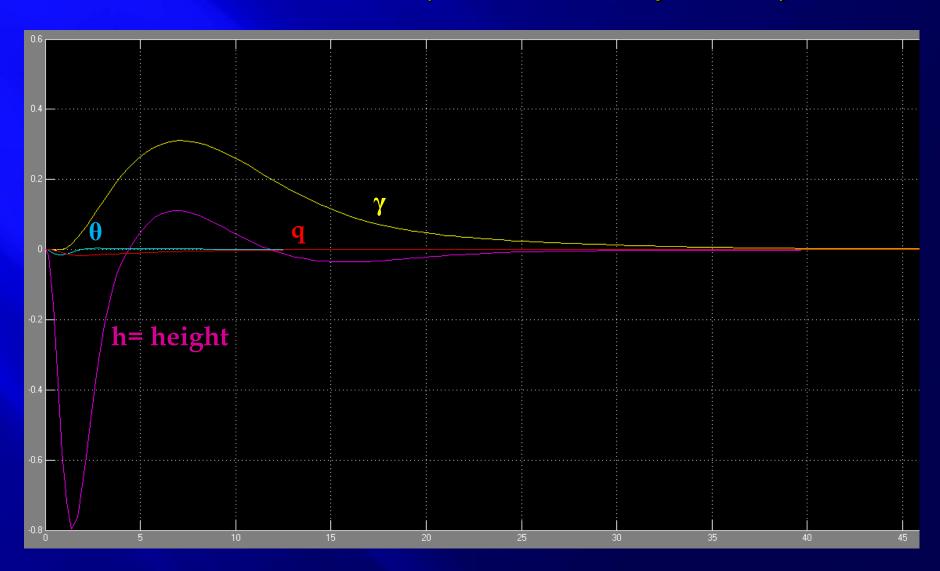


# **HEADING (Directional systems)**





### **HEIGHT HOLD (Directional systems)**





# **Aproach (Navigational systems)**



# ANY QUESTIONS?

